



Cophasing and beam combination

for IRSI-Darwin

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IRSI /Darwin



ESA's first space interferometry mission:

find/characterize earthlike planets,
AND
perform high resolution imaging

- six 1.5 m telescopes
- baseline range 50 m -> 500 m
(up to 4 mas resolution at $10\ \mu\text{m}$)
- science channel: 6 to 20+ micron
- intrasatellite metrology ($\sim 100\ \mu\text{m}$)
- mN and mN FEEP thrusters
- passive cooling $< 40\ \text{K}$

SRON is active in IRSI /DARWIN since early 2000. Focus on the **imaging** component, together with Leiden Observatory

- * Critical review of Alcatel imaging study

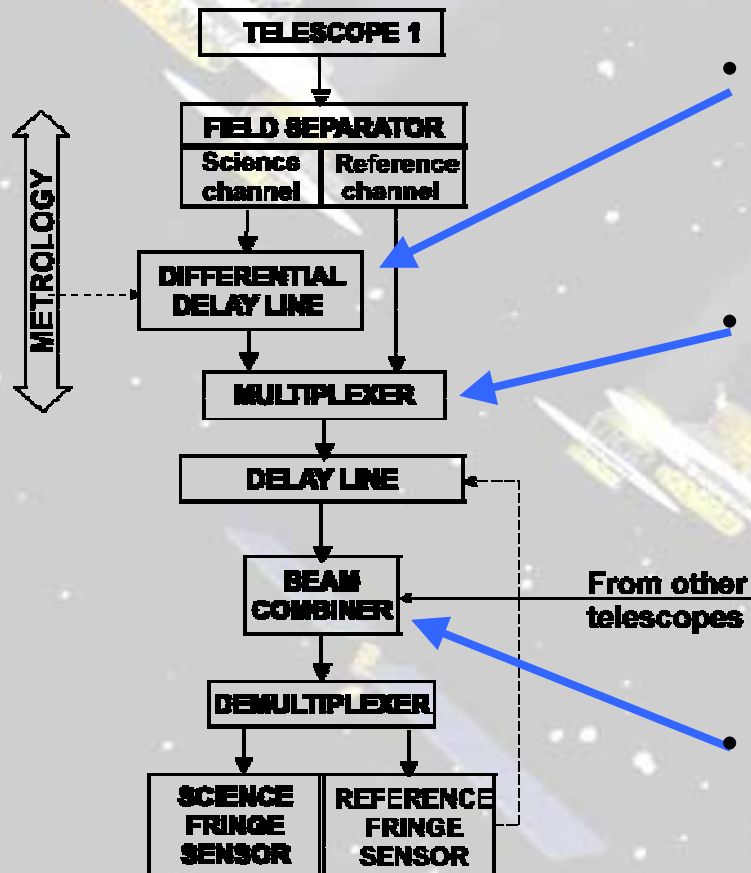
- * PhD student from november 2001 working on:

 - * end-to-end modelling (τ_0 on axial/science channels)

 - * hardware validation (wide-field beam combiner within the TPD/TUD Dutch Testbed Interferometer)

- **SMART2** (DARWIN precursor, fringe tracking from star?)

Critical review of Alcatel imaging study



- Assessment of additional metrology needs driven by imaging
- Comparison of available multiplexing techniques
- Six-way beam combiner for imaging

Blind tracking of external OPD evolution $OPD_x = \vec{B} \vec{\theta}$

Compensate blindly the evolution of OPD_x with DDL.

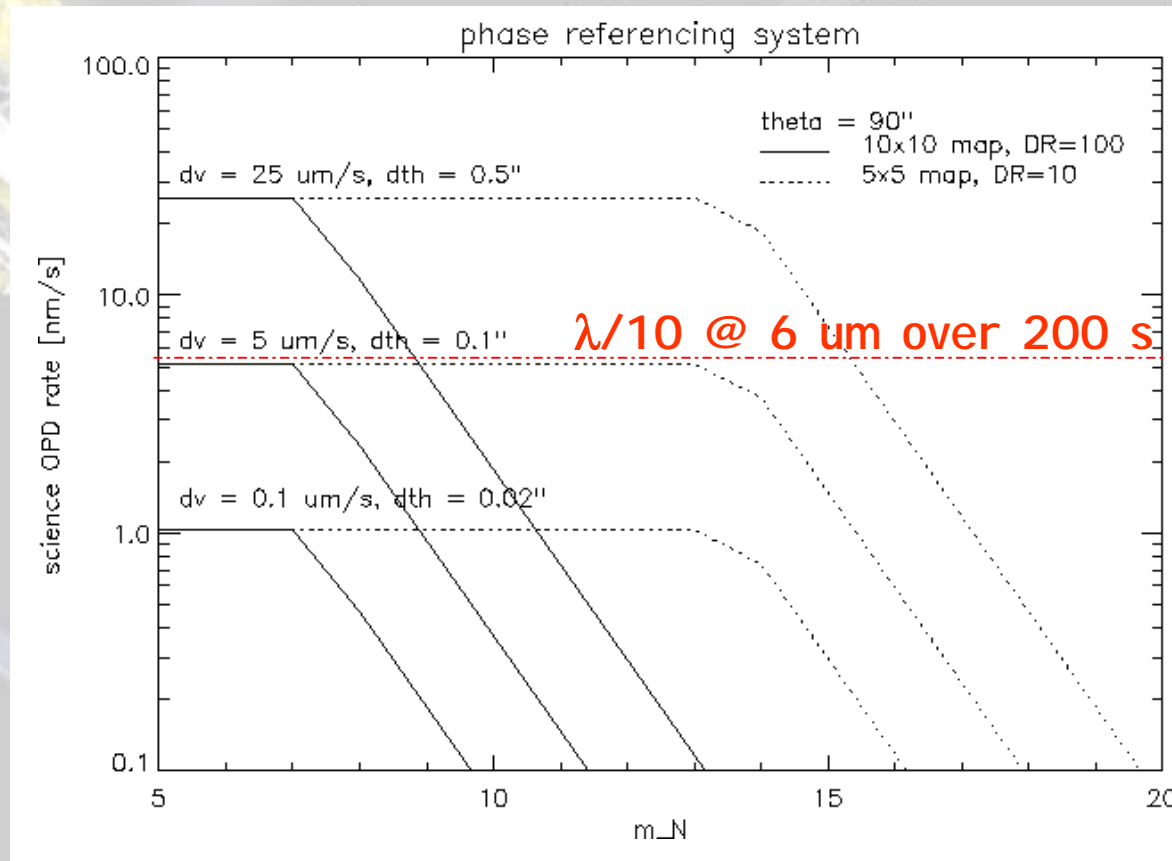
Tracking requirement: < 600 nm OPD change within one coherence time T_{cl} (~ 200 s)

Derived measurement requirements:

* Star separation, to $d\theta < 0.12''$ (\sim PSF/16)

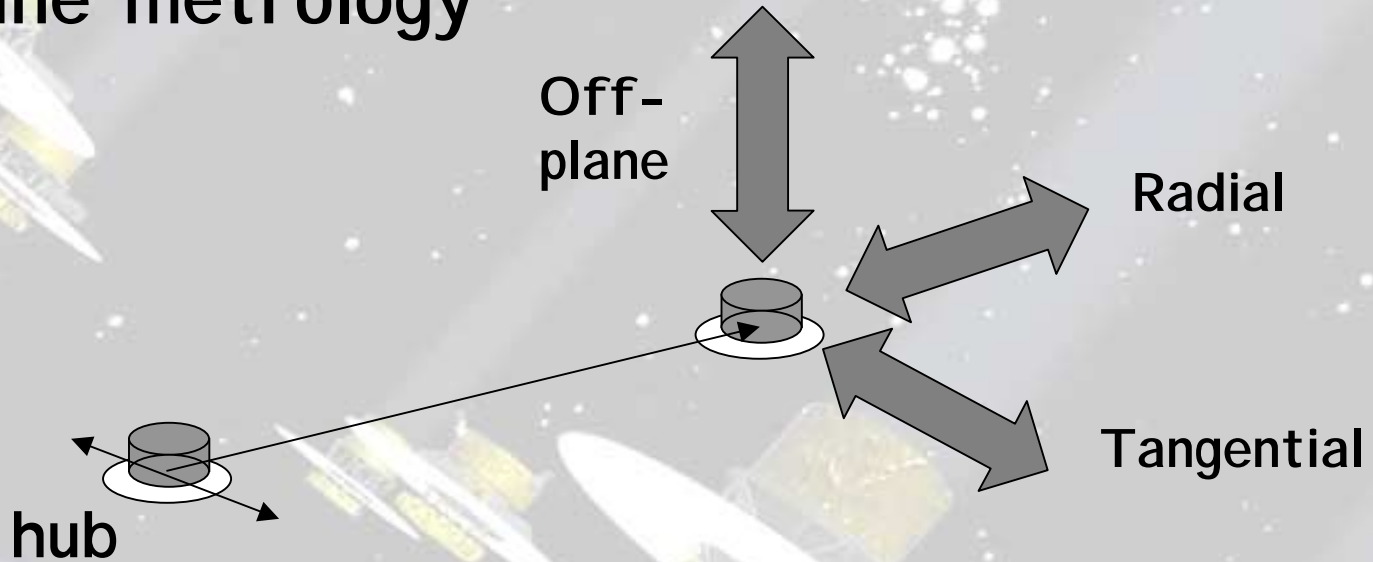
* baseline rates, to $dv < 7 \mu\text{m/s}$

(assuming a measurement rate of $1/T_{cl}$)



dV = error in knowledge baseline rate
dth = error in knowledge of θ

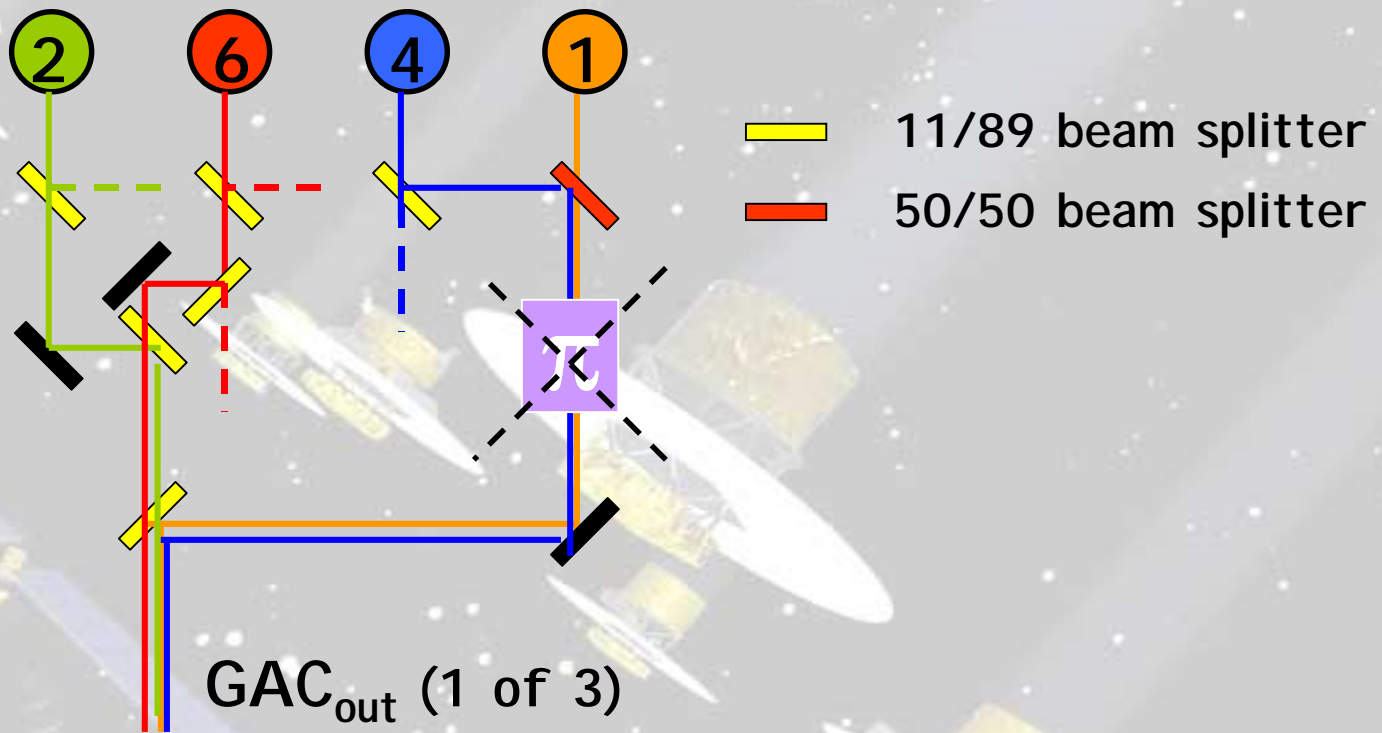
Baseline metrology



| | | | |
|-------------|----------------------------|--------------------------|------|
| Length: | laser metrology | 0.1 mm rms @ 1 Hz | OK! |
| Off-plane: | APS sensor | 5 mm/ 25 μ m @ 10 Hz | OK! |
| Tangential: | RF goniometry | B tan(0.1°) @ 10 Hz | no.. |
| | complement with APS sensor | 5 mm/ 25 μ m @ 10 Hz | OK! |

Beam combination

Current design: use nulling BC (without π phase shifter) for both ref/science beams. 'Kind-of' all-in-one pupil-plane BC.



$$GAC_{out} = \frac{1}{2} A_1 + \frac{1}{6} A_4 + \frac{1}{3} A_2 + \frac{1}{3} A_6$$

pros • minimizes added complexity for imaging

cons • small field of view $\theta = 0.3''$, $dOPD/d\theta \sim 2.5 \text{ mm/arcsec}$
($\lambda = 10 \mu\text{m}$, $R = 300$, $dV/V < 0.1$)

- 10 fringe outputs, unbalanced amplitudes
- temporal modulation imposes short integration times
- temporal modulation is **chromatic**

Alternative approach:

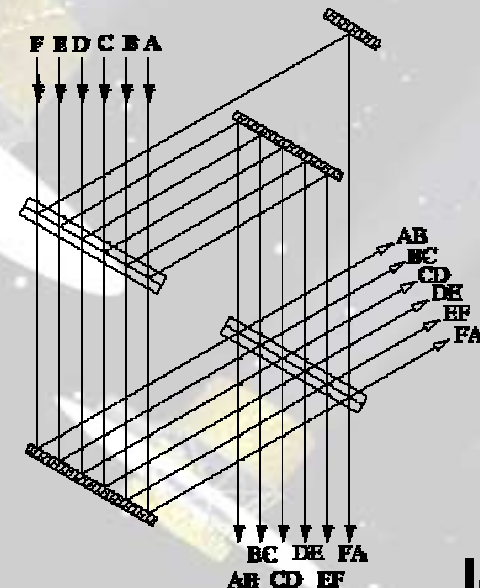
- separate BC spacecraft for imaging (increases redundancy)
- develop dedicated BC's for OPD tracking and science
- if possible, implement achromatic modulation

Reference (tracking) beam combiner

- Optical bandwidth : 1-2.3 μm
- Only five shortest B's needed
- Radial laser metrology ensures coherencing to 100 μm rms

A possible approach:

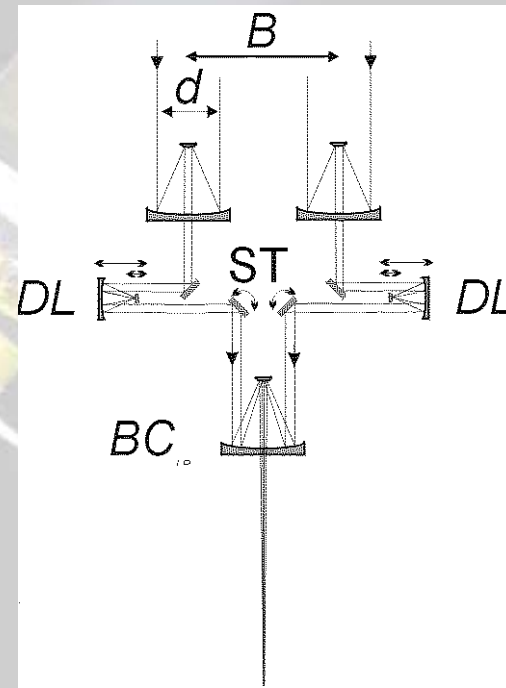
Partial pairwise scheme with 180°
achromatic spatial modulation
(no moving parts)



Science beam combiner:

- Optical bandwidth : 6-20+ μm
- Visibility measurements on all fifteen baselines
- wide field. Goal: 3xPSF's cophased to $\lambda/4$ @ 6 μm

A reasonably cophased field of view can be only achieved with an image-plane beam combiner with homothetic pupil mapping



Main requirements

- Pupil positioning accuracy: 36 μm rms over ~ 0.5 m stroke
- Calibration of imaging parameters
 - image matching reqs.: $dM/M < 10^{-2}$, $d\phi < 2$ degrees
 - pupil mapping reqs.: $dM/M < 1.8 \times 10^{-5}$, $d\phi < 3.6''$
- Optics quality: 450 nm rms in the field for whole optical train
- Focal plane: $\sim 2\text{k} \times 2\text{k}$ detector, image zooming

Main challenges:

- accurate pupil positioning at cryogenic temperatures
- spectrometer implementation: FTS?
- cryogenic active optics (fine-tuning optical trains, zooming)

Conclusions

Phase referencing

- need for extra array metrology (duplicate existing subsystem)
- validate multiplexing concept

Beam combiner:

- Define focal plane implementation
- Initiate technical studies for positioning and active optics devices
- Deepen understanding of optics/alignment/calibration requirements, iterate on fov goal

Thrust-limited baseline rates

